D'Shannon Products, LTD 1309 County Road 134 Buffalo, MN 55313 Document No: FMS-DP-KM35 TT Hawker Beechcraft 135, M35

FAA APPROVED

PILOT'S OPERATING HANDBOOK AND

FAA APPROVED AIRPLANE FLIGHT MANUAL SUPPLEMENT

FOR

HAWKER BEECHCRAFT MODEL K35, M35 (s/n D-5726 through D-5564)

NORMAL CATEGORY

(Operation in excess of 2950 lb. Max. Gross Weight, or with Feel in Tip Tanks)

UTILITY OATEGORY

(Operation at 2950 lb. Max. Gross Weight or Less - Tip Tanks Epopty)

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SER. NO

This supplement must be attached to the Pilot's Operating Handbook and FAA Approved Airplane Flight Manual when two 20 gallon auxiliary wing tip fuel tanks are installed in accordance with STC(s) SA153EA or SA0272CH. The information contained herein supplements or supersedes the basic handbook only in those areas listed herein. For limitations, procedures, and performance information not contained in this supplement, consult the basic Pilot's Operating Handbook and FAA Approved Airplane Flight Manual.

FAA APPROVED:

Charles L. Smalley, Manager Chicago Aircraft Certification Office Federal Aviation Administration Department of Transportation Federal Aviation Administration Des Plaines, IL 60018

SECTION I GENERAL

This supplement contains revised information for the basic airplane when operated in accordance with STC(S) SA153EA or SA02722CH. The information contained herein supplements or supersedes the basic handbook only in those areas listed herein. Consult the Pilot's Operating Handbook and FAA Approved Flight Manual for limitations, procedures, and performance information not contained herein.

MAXIMON SERTI IED WEISTI	MAXIMUM	CERTIFIED	WEIGHT
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Maximum Ramp Weight..... Maximum Take-off Weight Maximum Landing Weight

SECTION II LIMITATIONS

GENERAL

The Airplane Flight Manual for this airplane ists information for operation in the UTILITY category. Since the tip tank installation is approved contingent on operation of the airplane in the NORMAL category when operated in excess of 2950 lb. of with fuel in Tip Tanks, the following Limitations supersede those of the basic Airplane Fligh Manual.

This airplane is eligible for operation in accordance with STC(S) SA153EA or SA02722CH and this airplane flight manual supplement only when equipped with the following modifications:

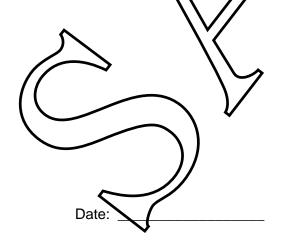
a) Wing Tip Fuel Tanks (STC(S) SA153EA or SA02722CH)

AIRSPEED LIMITATION

Maneuvering Speed,CAS 118 knots CAS 136 mph

WEIGHT LIMITS

Maximum Ramp Weight.....3160 lb. Maximum Take-off Weight Maximum Landing Weight.3150 lb.



CENTER OF GRAVITY LIMITS (Landing Gear Extended)

FORWARD LIMITS

77.0 inches aft of datum to 2475 lbs. with straight line variation to 82.1 inches at 3150 lbs.

AFT LIMITS

84.7 inches aft of datum at all weights.

MANEUVER LIMITS

This is a NORMAL CATEGORY airplane when operated in excess of 2950 to or with fuel in Tip Tanks. Spins and acrobatic maneuvers are prohibited. Normal category sirplanes are limited to Non-acrobatic operation.

Non-acrobatic operation includes:

- 1. Any maneuver incident to normal hying.
- 2. Stalls (except whip stalls)
- 3. Lazy eights, chandelles, and steep turns, in which the angle of bank is not more than 60°.

Spins are prohibited.

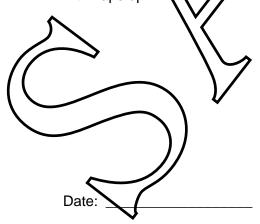
No inverted maneuvers are approved

FLIGHT LOAD FACTORS

FUEL

In addition to the basic airplane fuel system, two auxiliary wing tip fuel transfer tanks are installed with a capacity of 20 gallons each, all of which is usable.

Take-offs are prohibited with more than 1/4 difference in tip tank fuel quantity. During flight if tip tank fuel quantity gauges indicate more than 1/2 tank difference the landing should be made with flaps up.



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PLACARDS

In Full View of Pilot:

FUEL CONSUMPTION MAY EXCEED TIP TANK TRANSFER RATE. INITIATE TRANSFER WITH BOTH MAINS AT LEAST ½ FULL. MONITOR MAIN TANK GAUGES TO PREVENT OVERFLOW. TRANSFER TIP TANK FUEL IN LEVEL FLIGHT ONLY.

In Full View of Pilot (Airspeed values are CAS)

NORMAL CATEGORY AIRPLANE

(WHEN OPERATED IN EXCESS OF 2950 LB. MAX. GROSS WEIGHT, OR WITH FUEL IN TO TANKS)

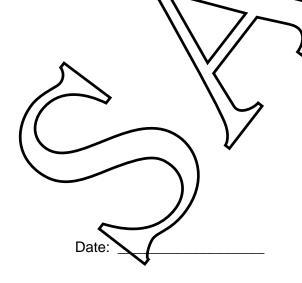
AIRSPEED LIMITATION (NORMAL CAT. OPERATIONS)

MAXIMUM DESIGN MANEUVERING SPEED 136 MPH (118 KNOTS)

OPERATE IN ACCORDANCE WITH FAM APPROVED FLIGHT MANUAL / PILOTS OPERATING HAMDBOOK. INTENTIONAL SPINS ARE PROHIBITED. NO ACROBATIC MANEUVERS APPROVED.

SECTION III EMERGENCY PROCEDURES

If for any reason it is necessary to land with more than 1/2 tank difference in tip tank quantities, the landing should be made with wing flaps in the "up" position.



SECTION IV NORMAL PROCEDURES

AIRSPEEDS FOR SAFE OPERATION

Maximum Turbulent Air Penetration

AS 120 KTS

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PREFLIGHT INSPECTION

Fuel drains are located on the lower surface of each tip tank. Orbin these points daily before the first flight to purge any water from the system.

Check security of flush mounted tip tank filler caps during preflight inspection.

Before flight, check the tip tanks for unsymmetrical fuel loading. If fuel tank capacities differ more than 1/4 tank, relocate fuel prior to take-off

See Section 7, Systems for additional information.

SECTION V PERFORMANCE

The performance listed in the basic Airplane Flight Manual is applicable to this airplane with the tip tank installation at the gross weight listed in the basic Airplane Flight Manual. Since the certification basis of the tip tank installation does not include a requirement that performance be made available in the AIM, and since the modifier did not choose to supply this information, no performance is listed at gross weights above that of the basic airplane.

SECTION VI WEIGHT AND BALANCE Weight Ce Limit Condition 82.1 3150 lb. 84.7 77 84.7 5 lb. or les CG Limitation wheels down 3150 Gross Note: Weight CG of Tip Tank Fuel 2475 Is 87" Aft of Datum 82.1 84.7 Inches Aft of Datum

Weight and Balance Loading Form

Model	Date:

Serial No: D- _____ Reg. No.:

ltem	Weight	Mom./100
1. Basic Empty Weight		1
2. Front Seat Occupants		
3. 3 rd and 4 th Seat Occupants	4	
4. 5 th and 6 th Seat Occupants		
5. Baggage		
6. Cargo)/
7. Sub Total Zero Fuel Condition		
8. Basic Fuel Loading		
9. Tip Tank Fuel Loading		
10. Sub Total Ramp Condition		
11. Less Fuel for Start, Taxi, and Take-o		7
12. Sub Total Take-off Condition		
13. Less Fuel to Destination	7/ //	
14. Landing Condition	7	

^{*} Fuel for start, taxi, and take-off is normally 16 b.

Usable tip tank fuel is located at an average arm of 87 inches aft datum.

SECTION VII SYSTEMS DESCRIPTION

FUEL

In addition to the basic airplane fuel system, two auxiliary wing tip fuel transfer tanks are installed with a capacity of 20 gallons each, all of which is usable. Take-offs are prohibited with more than 1/4 difference in tip tank fuel quantity. During flight if tip tank fuel quantity gauges indicate more than 1/2 tank difference the landing should be made with flaps up.

Tip tank fuel is transferred into its respective main tank by an electric pump at a rate of approximately 15 gallons per hour. The transfer pump and a sole oid valve are mounted in side the wheel well of each wing on the rib at wing station 66. At higher power settings, fuel consumption may exceed the fuel transfer rate to the main tank selected.

Tip tank transfer pump switches are located either on the face of the instrument panel or between the front seats on the partition assembly to ward of the main spar truss. The pump and solenoid valve circuit breaker is installed adjacent to the pump switches.

A fuel drain is provided on the lower surface of each tip tank

Fuel quantity is measured by observing the fuel level on a sight gauge located on the inboard side of each tip tank.

Normal tip tank fuel transfer should be accomplished simultaneously to maintain symmetrical wing tip tank fuel loading. Initiate transfer with the left main at 1/2 fall and feeding the engine. During the transfer, monitor fuel gauges for both main tanks and stop transfer if gauge indicates full to prevent overflow of fuel through the main tank vent tubes.

SECTION VIII HANDLING, SERVICING AND MAINTENANCE

No Change.

SECTION IX SUPPLEMENTS

No Change.

SECTION X SAFETY INFORMATION

No Change.

